

# Navigation Systems Clocks Technologies

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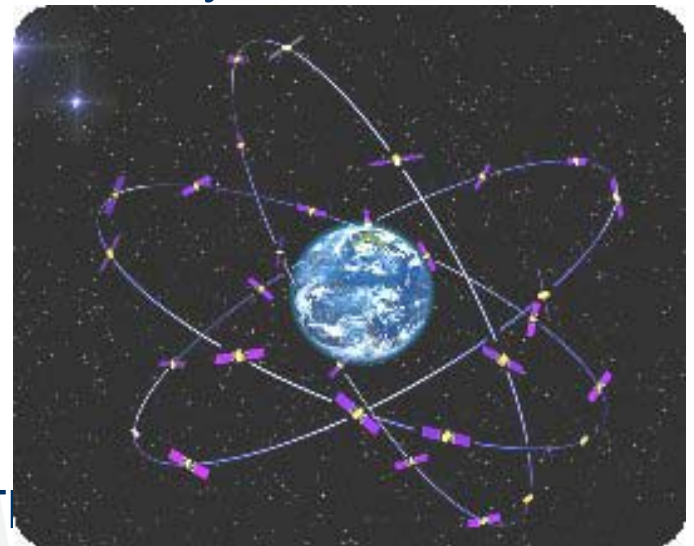
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# Presentation Outline

1. Development Steps , Performances and Status of the Galileo Space Clocks
2. Evaluation of the accuracy of the Space Clocks taking into account measurement noise induced by geodetic technique in measuring and predicting the Space Clocks
3. On board Galileo Time Keeping
4. On ground system clock generation
5. Conclusions

# Galileo program description

- Global Navigation Satellite System Jointed by EC & ESA
- Constellation
  - 30 satellites
  - 3 circular MEOs
  - Altitude 23222km
  - Inclination 56°
- Mission life time of 12 years
- Clock model update < 10'000 sec.(T)
- Metric/sub-metric navigation accuracy -> 1 to 3 ns clock model error



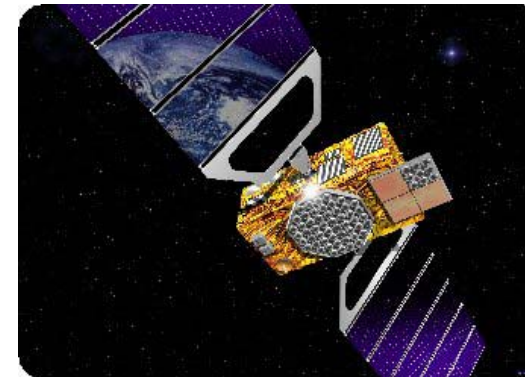
# Onboard Clocks for Galileo Development History

- Two baseline clock technologies
  - Rubidium Atomic Frequency Standard (RAFS)
  - Passive Hydrogen Maser (PHM)
- Long heritage for both clocks in Switzerland since '80s
- Continuous support from ESA & Swiss Space Office especially since set-up of the GNSS2 program

# In-progress GSTB-V2

## First Flight Opportunity for Galileo Clocks Validation !

- Two experimental satellites due for launch end of 2005 & 2006 for:
  - Frequency Filling
  - Test of critical technologies (like clocks)
  - Experimentation on Galileo Signals
  - Characterisation of MEO environment
  
- Navigation Payload embarking
  - One PHM unit and two RAFS units (GIOVE-B)
  - Two RAFS unit (GIOVE-A)

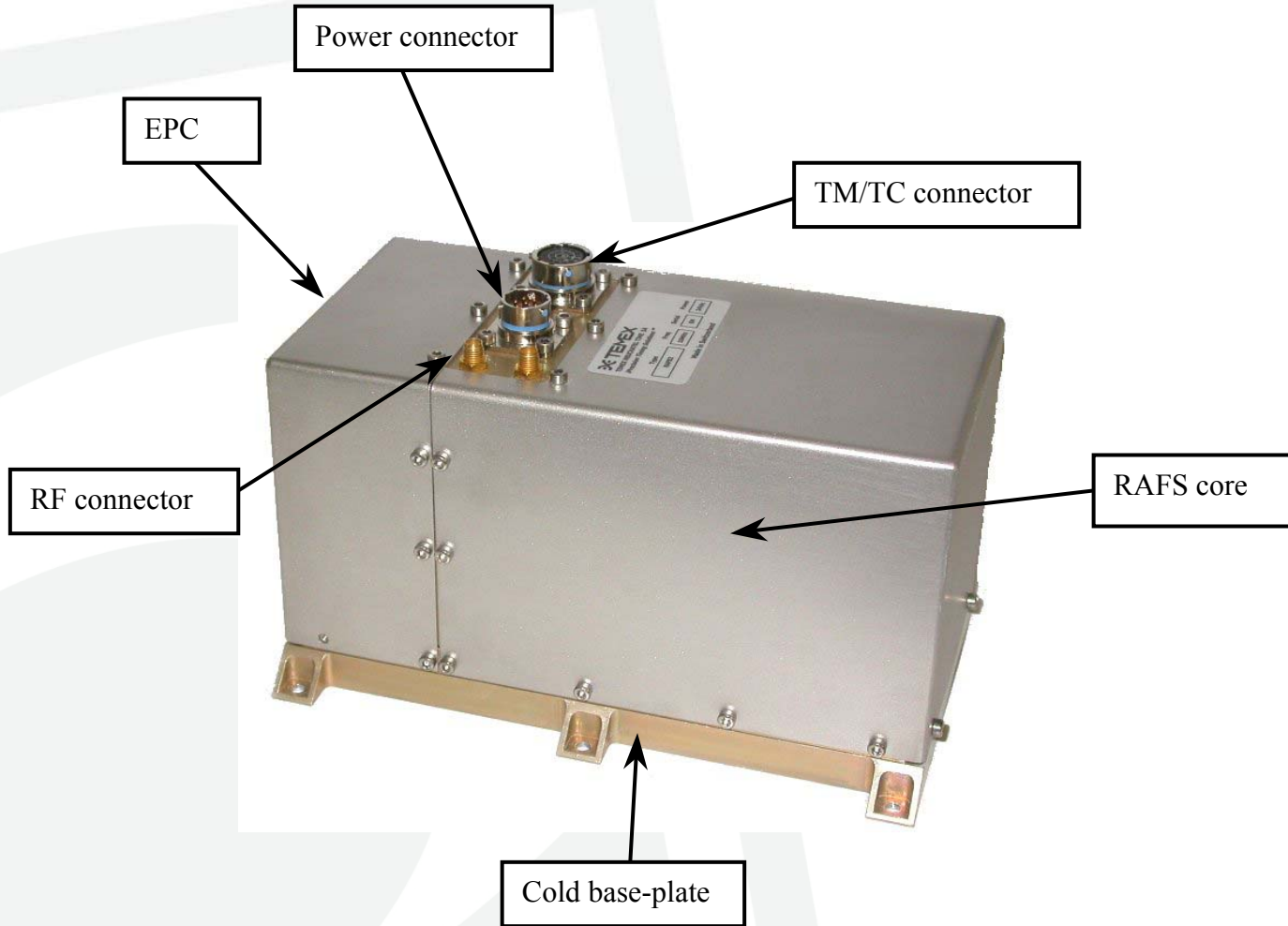




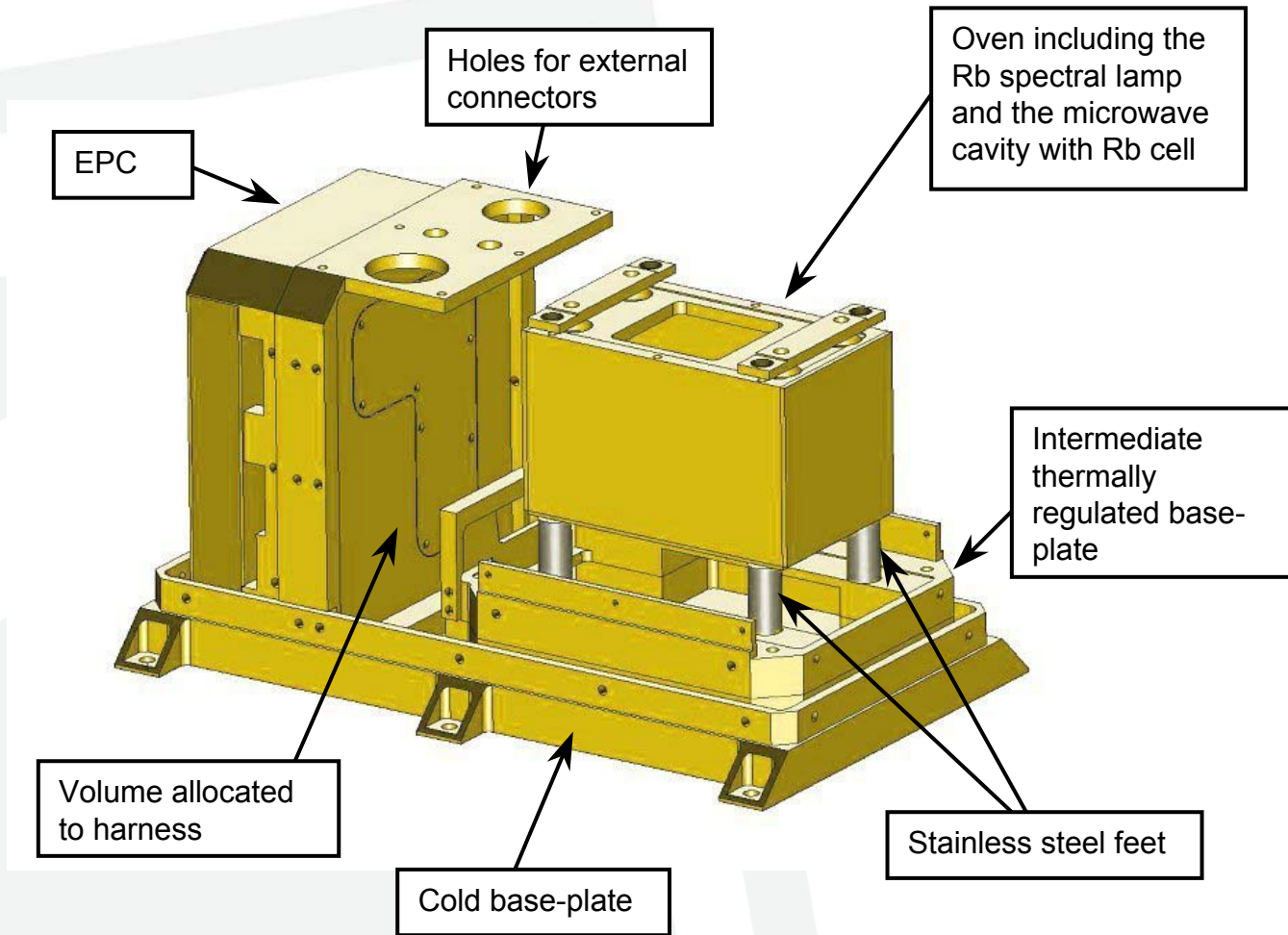
# Development Milestones

Steps	RAFS	SPHM	OPCs
B	completed in 1995	Based on VREMYA design +ON/GA analysis (1999)	$< 3E-12 * \tau^{-1/2}$ on gnd Cs OP tube in 2002
M	Completed in 2000	Completed by Q2 / 2003 by ON&GA	Will start in Thales in 2006
QMs	5 models in specs in 2002 Lifetime still on-going	Industrialisation and EQM Completed by Q4/2003 by GA&TNT	-----
M	Qual. Tests completed Rad test (CNES) Q1/2003	Qualifications in 2006	-----
Ms or FMs	6 models delivered for Giove A & Giove B Satellites in 2005	Three models delivered for Giove B in 2005 - 2006	-----

# RAFS for GIOVE\_A



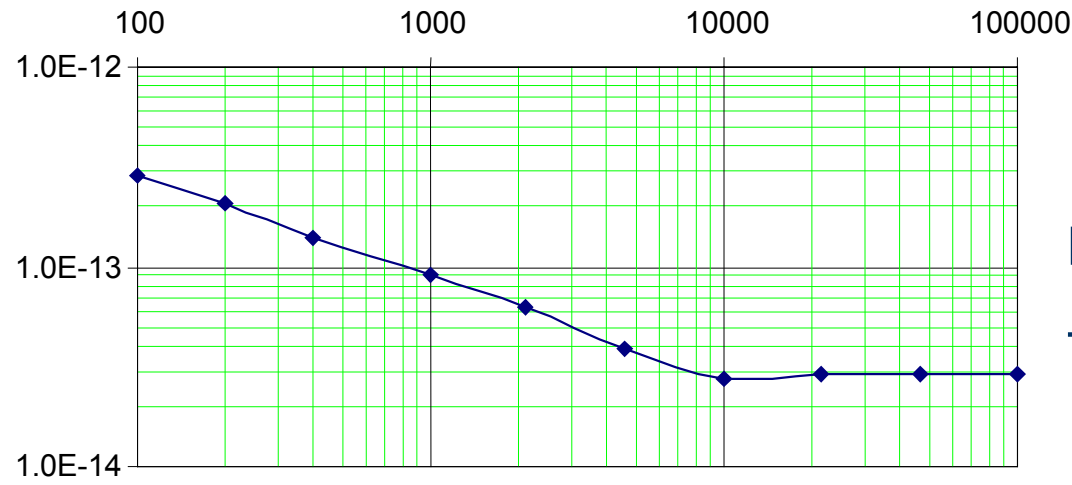
# RAFS internal construction



# Performance Achieved

## RAFS: Frequency & Time Stability

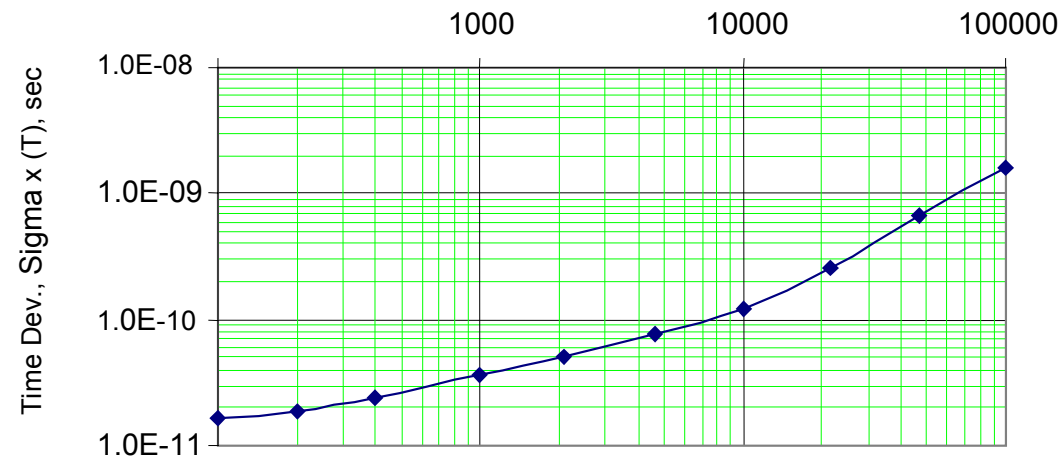
Averaging Time, sec.



Drift removed: 2E-13 / day

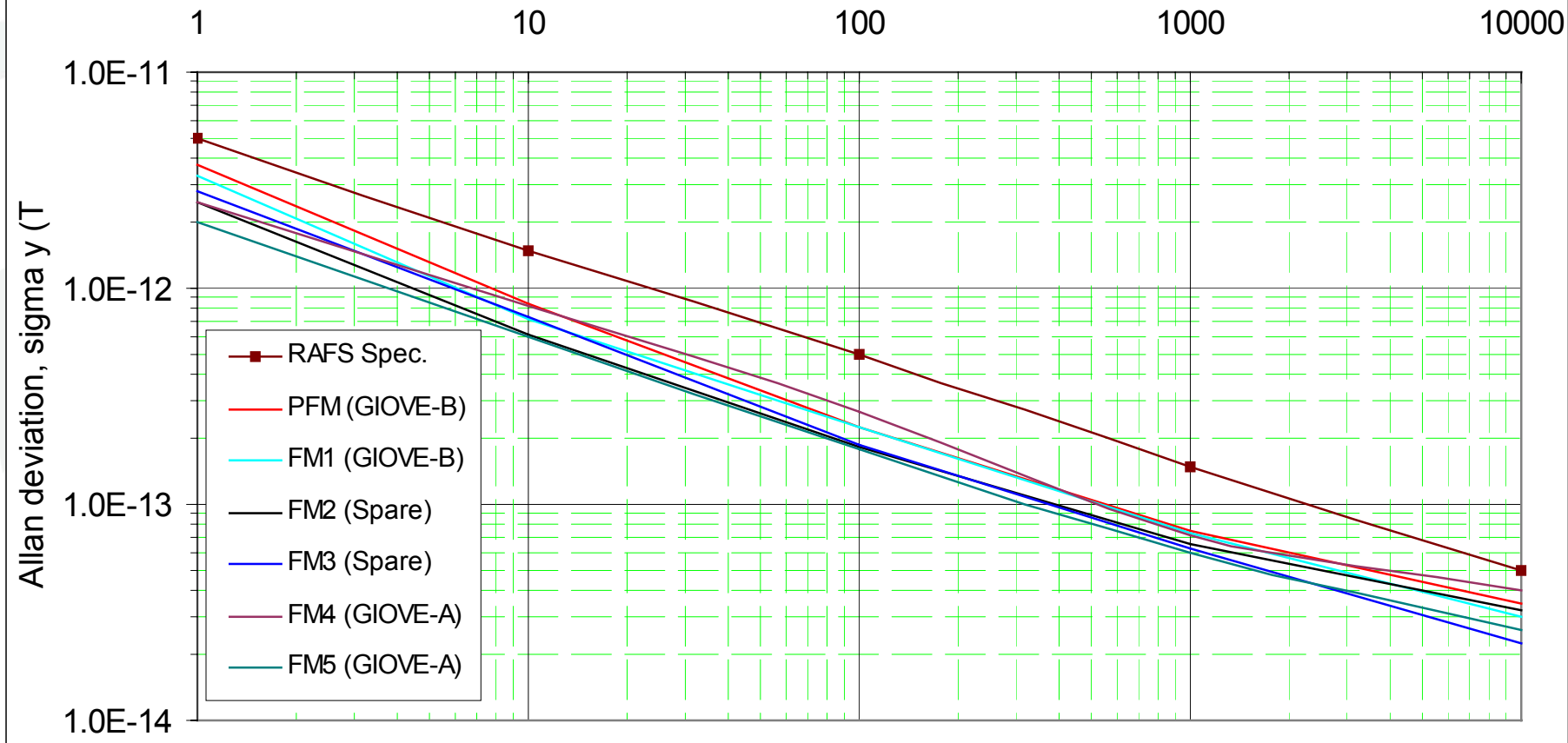
Temperature: +-1°C

Averaging Time, sec.



## Frequency stability of GSTB-V2 RAFS (Q4 2005) drift removed

Averaging time, T, in seconds



# RAFS2 for GSTB-V2 Performance Achieved

- ✓ **Frequency stability:**  $< 4 \times 10^{-14}$  @ 10'000 sec
- ✓ **Flicker floor :**  $< 3 \times 10^{-14}$  (drift removed)
- ✓ **Thermal sensitivity:**  $< 5 \times 10^{-14}$  /°C
- ✓ **Magnetic sensitivity:**  $< 1 \times 10^{-13}$  / Gauss
- ✓ **Mass and volume:** 3.2 kg and 2.4 lt

# **GIOVE-A Preliminary information (FM4 and FM5)**

- Start:** Several start sequences all 100% nominal
- Rb Light TM:** Nominal values and perfectly stable
- Rb Signal TM:** Nominal values and perfectly stable
- Operating Temp.:** Nominal
- Power:** Nominal

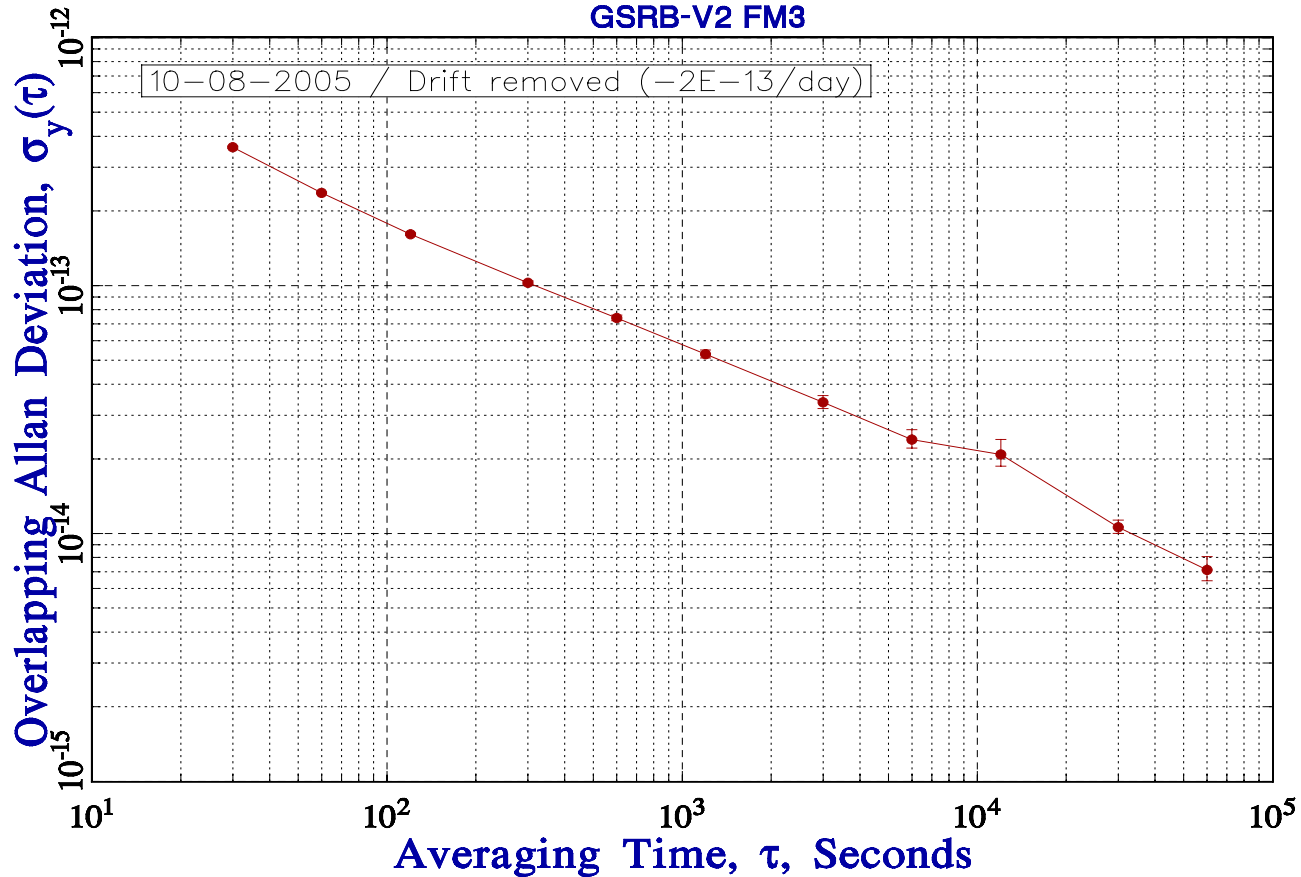
**Ground segment measurements showing almost same stabilities values than on ground .**

# RAFS3 further improvements

## Goal < 1 E-14 / day

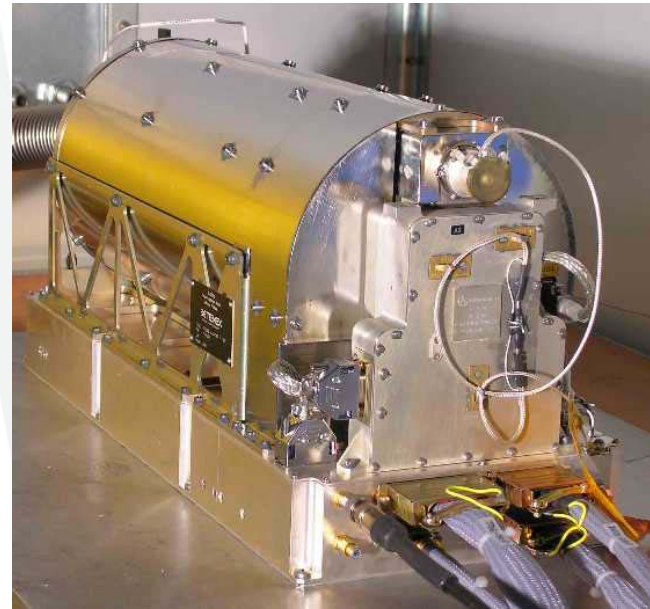
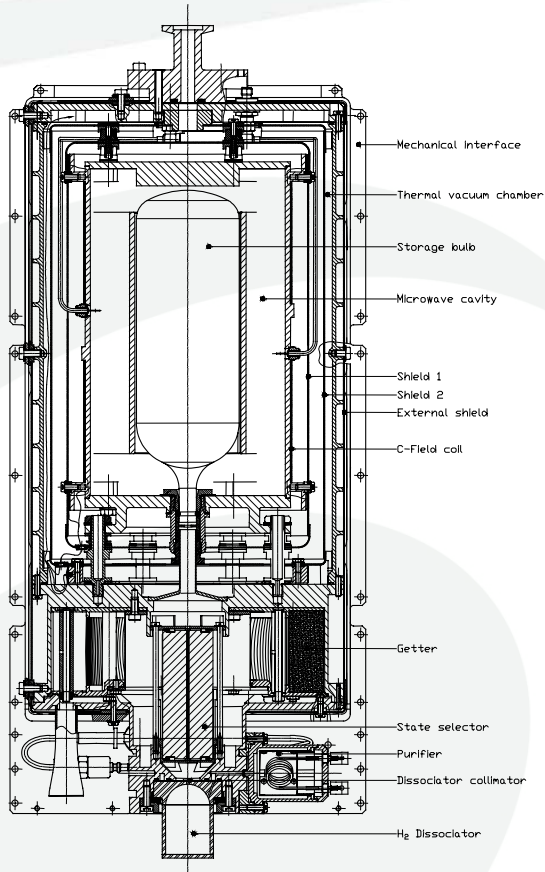
### FREQUENCY STABILITY

GSRB-V2 FM3



Stable32

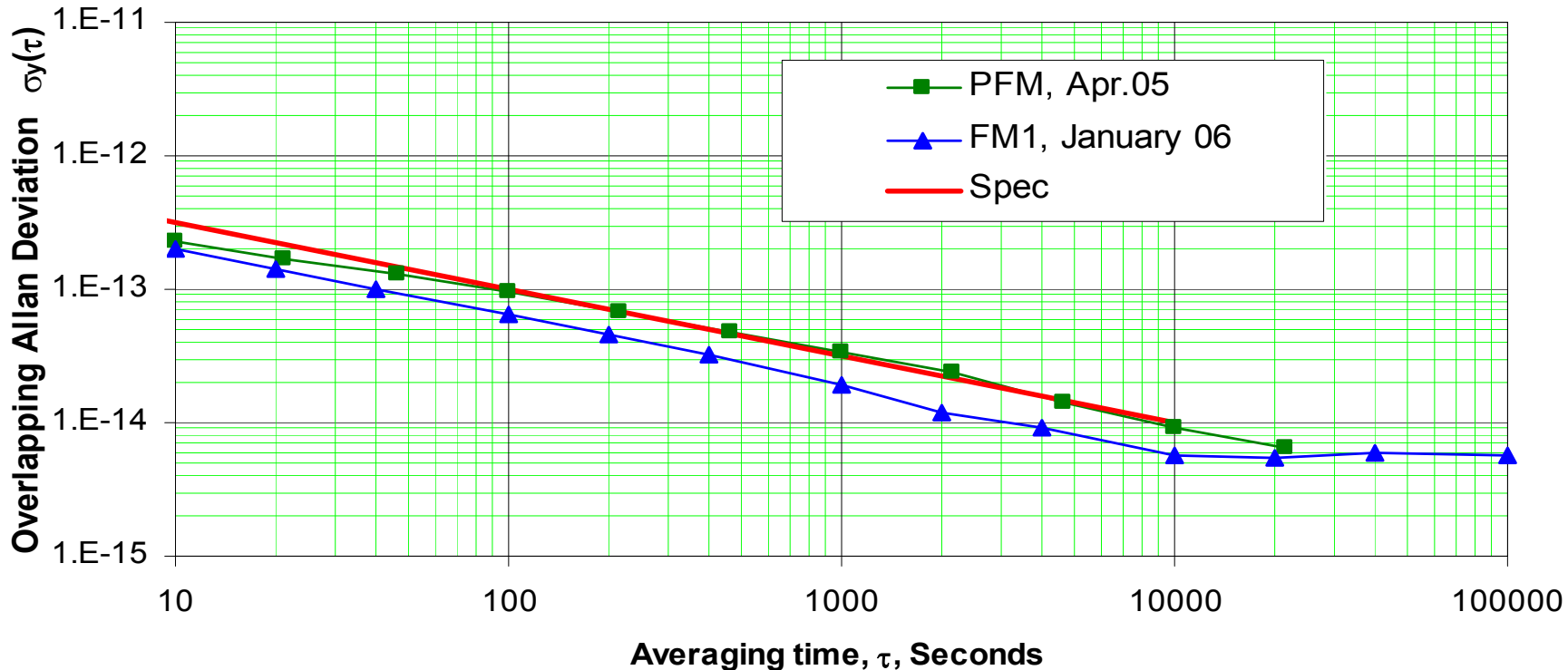
# Passive Hydrogen Maser for Galileo



Complete clock delivered for Giove B sat



# PHM/FM Performance Improvement



Significant improvements obtained between PFM and FM models by improving cavity quality factor

Drift:  $<5E-15$ /day

# PHM for GSTB-V2

## Performance Achieved

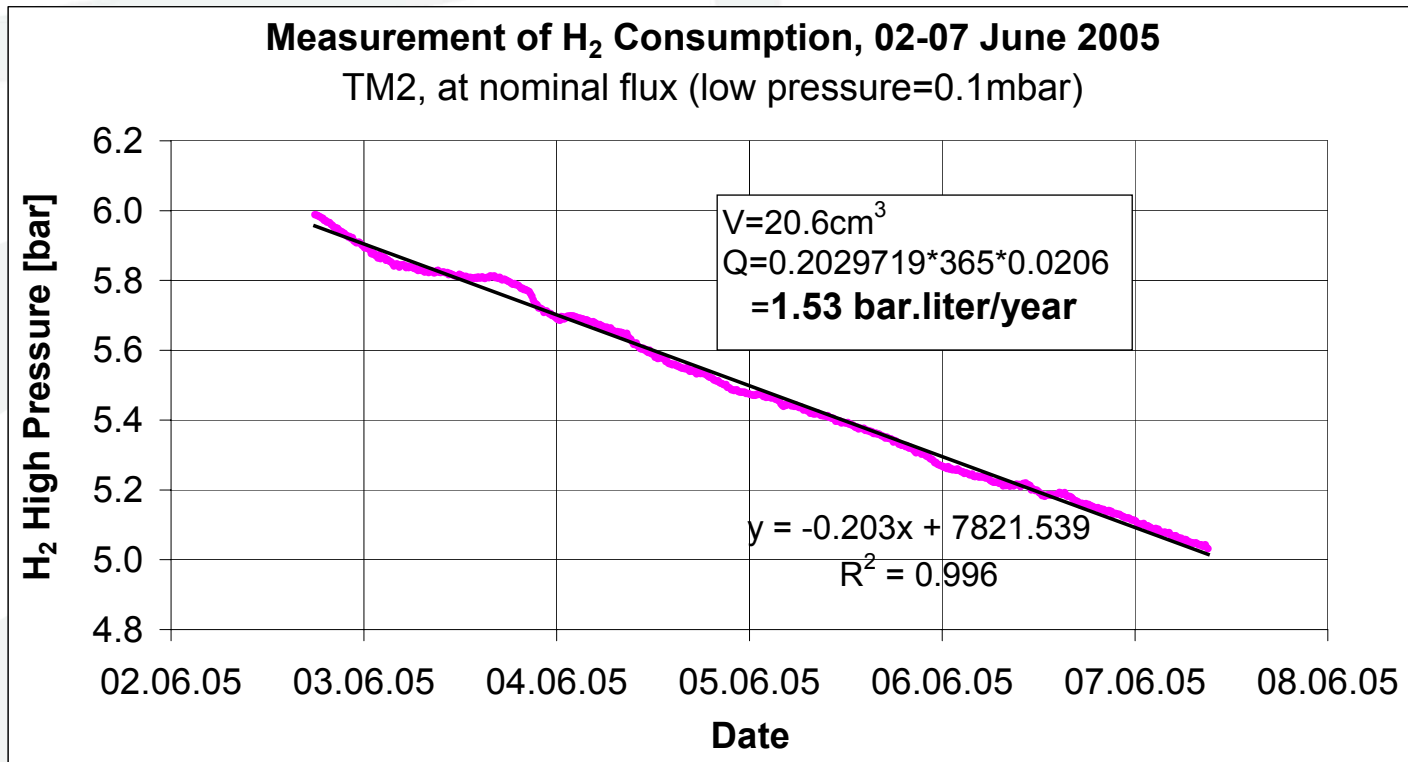
- ✓ **Frequency stability:**  $< 7 \times 10^{-13} \times t^{-1/2}$   
( $1\text{s} < t < 10'000\text{s}$ )
- ✓ **Flicker floor :**  $< 6 \times 10^{-15}$
- ✓ **Drift:**  $< 5 \times 10^{-15} / \text{day}$
- ✓ **Thermal sensitivity:**  $< 3 \times 10^{-14} / ^\circ\text{C}$
- ✓ **Magnetic sensitivity:**  $< 4 \times 10^{-14} / \text{Gauss}$
- ✓ **Mass and volume:** 18 kg and 28 lt

# PHM Life Time

- 12 years orbit life + 1 year ground storage
  - The orbit life is limited by
    - Size of hydrogen container
    - Bulk getters mass (hydrogen sorption capacity)
    - Ion pump sorption capacity (for background gases)
    - Total dose of ionising radiation

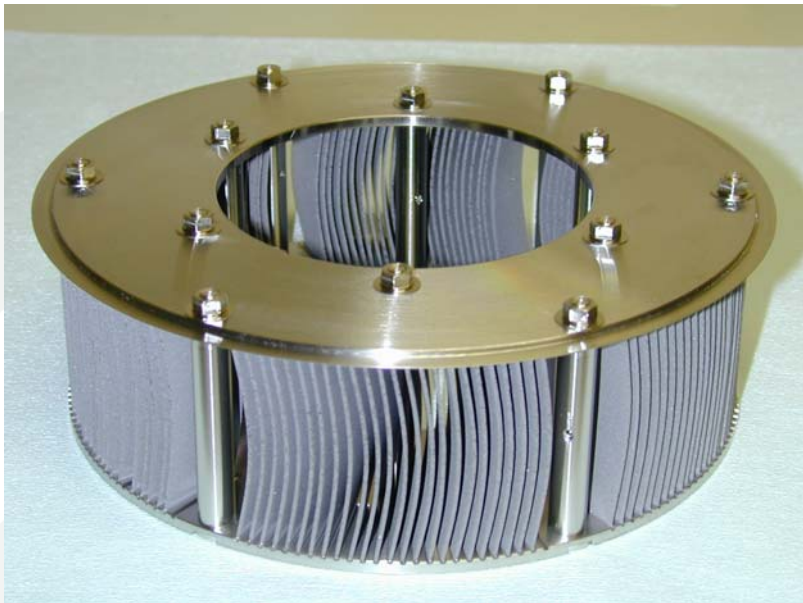
# PHM Life Time Factor 1: $H_2$ Container

## $H_2$ Consumption test



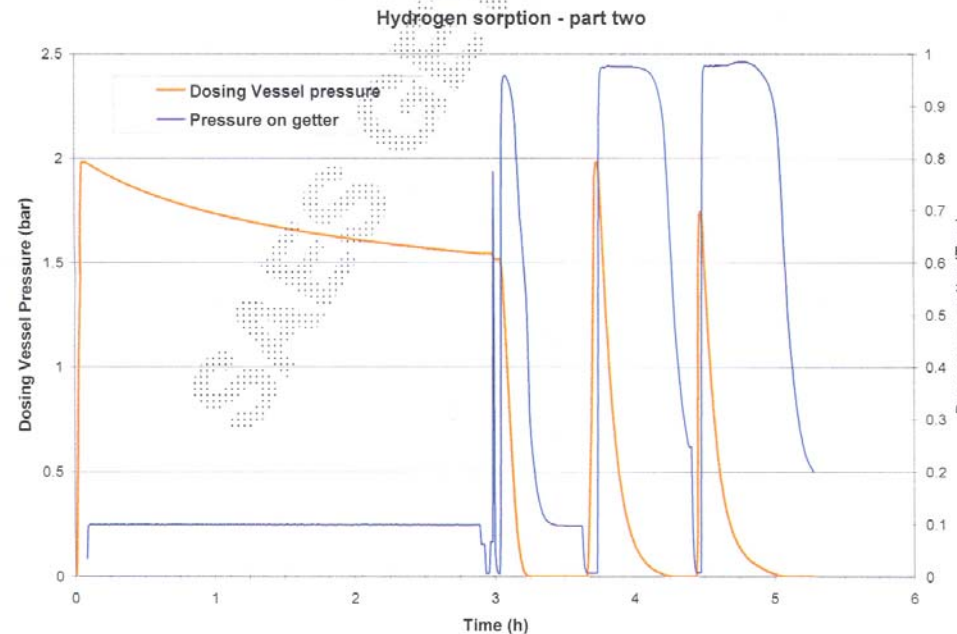
- $H_2$  container with the capacity of 25 bar.l is sufficient for 12-years life time, taking account of the margin.

# PHM getters & high vacuum chamber



## H<sub>2</sub> Sorption Test on Getter Cartridge

- Capable of sorbing the required amount of H<sub>2</sub> of 20 bar\**l*.
- Base pressure after sorption was in the low e-7 mbar range with only the getter cartridge pumping.



Sep. '03 at SAES Getter S.p.A

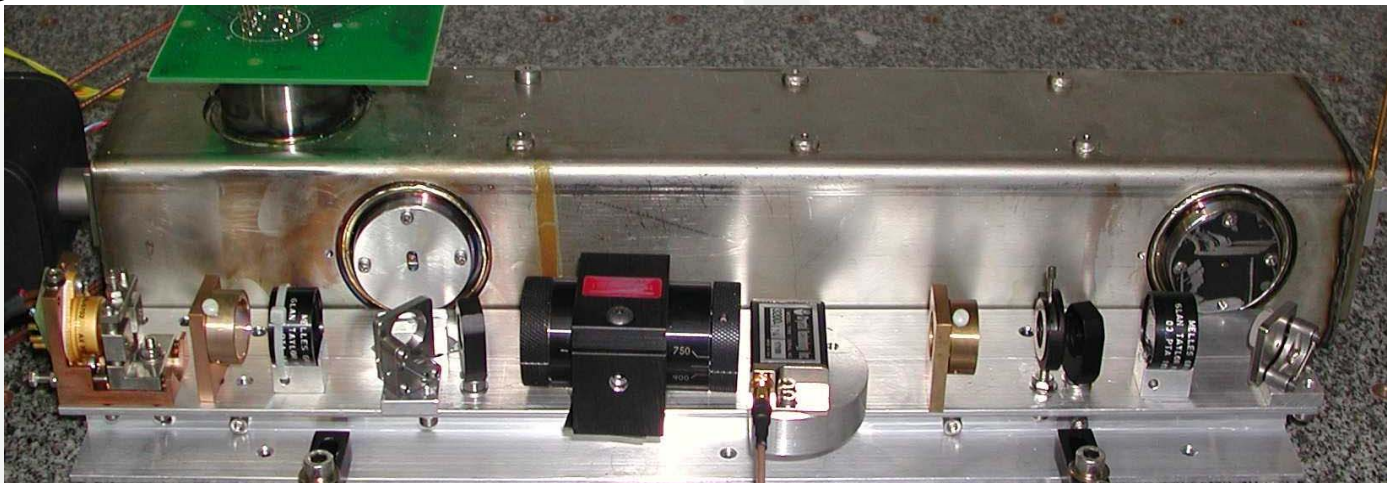
# PHM for GSTB-V2 Qualification Status

- ✓ **Vibration: Ok**
- ✓ **Shock: Ok**
- ✓ **EMC/EMI: Ok**
- ✓ **Thermal Vacuum: Ok**

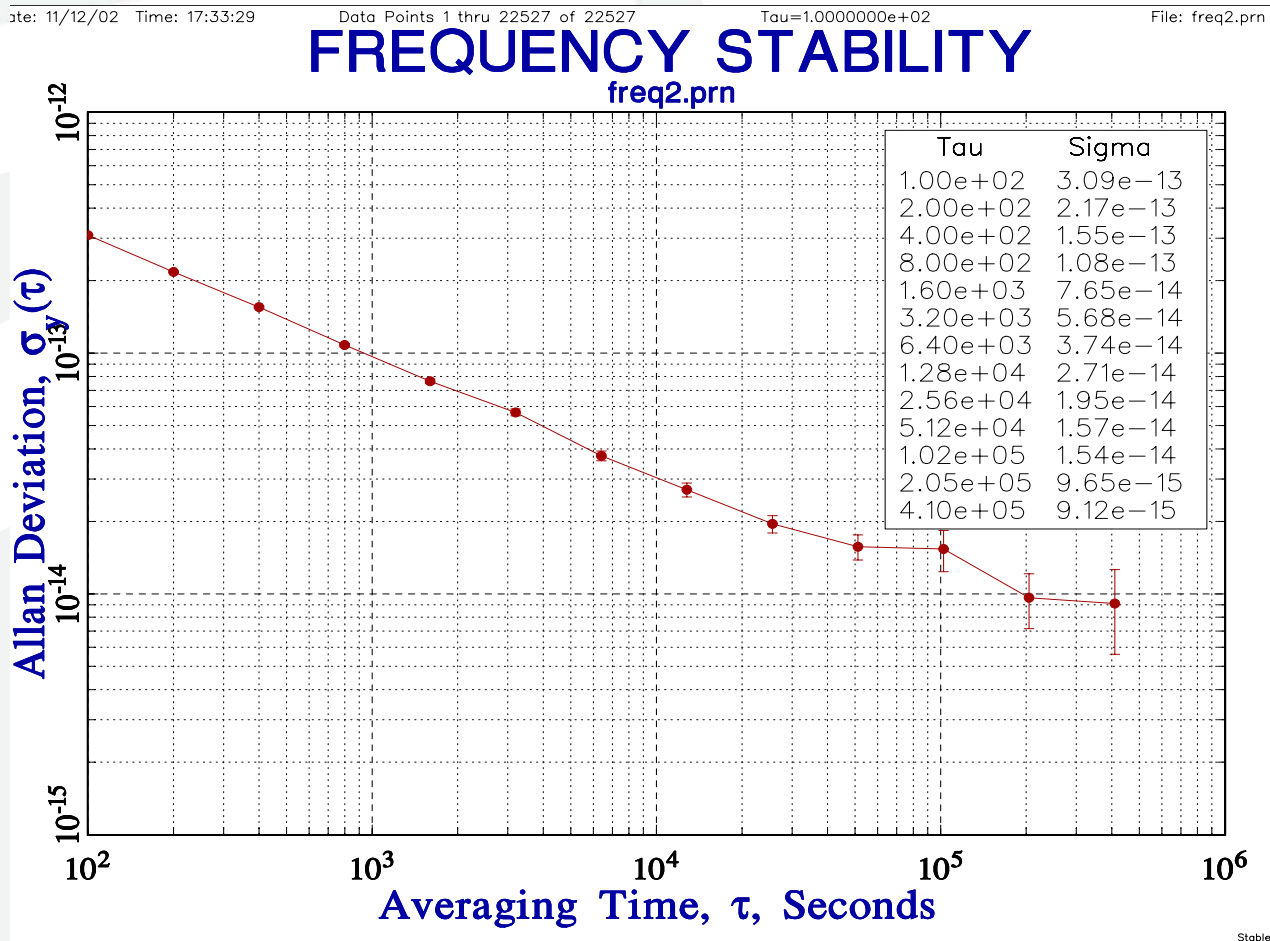


## Optically Pumped Cesium: BB (TEK SYS / 2002)

- The compact optically pumped cesium beam clock developed by Tekelec Systemes with the scientific support of the SYRTE exhibits a frequency stability of  $3 \cdot 10^{-12} \tau^{-1/2}$  measured against a hydrogen maser as a reference in 2002.
- The sealed tube works under a low atomic flux consistent with lifetime of 8 years or more. It uses a 22 cm long Ramsey cavity
- The compact 3 cm wide optical bench is rigidly fixed along the tube which length is 45 cm.

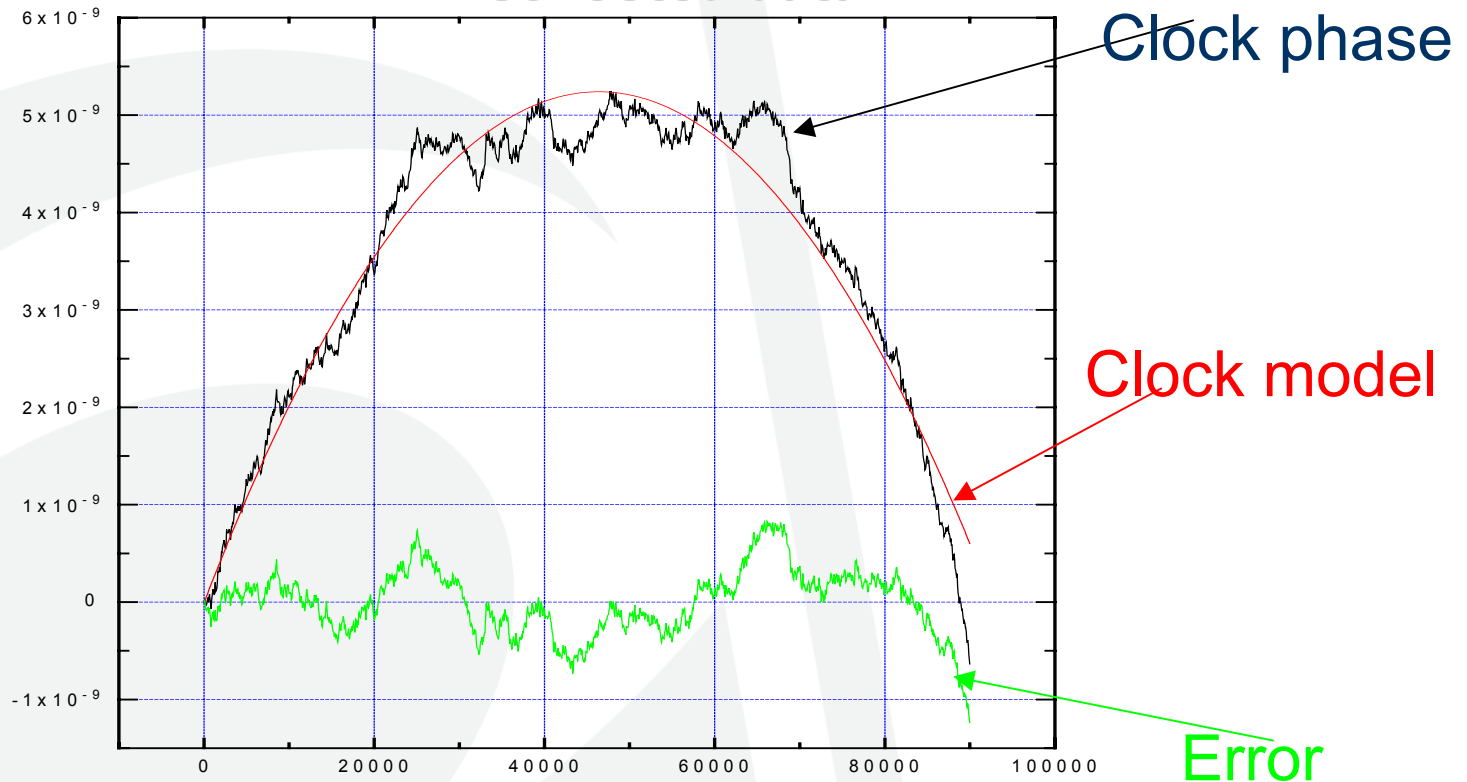


# OPCs :BB Stability (TEK SYS/ dec 2002)



# Clock model & prediction time error

**predictive system : error will be the difference between the clock phase-time & the establish clock model from previews collected data.**

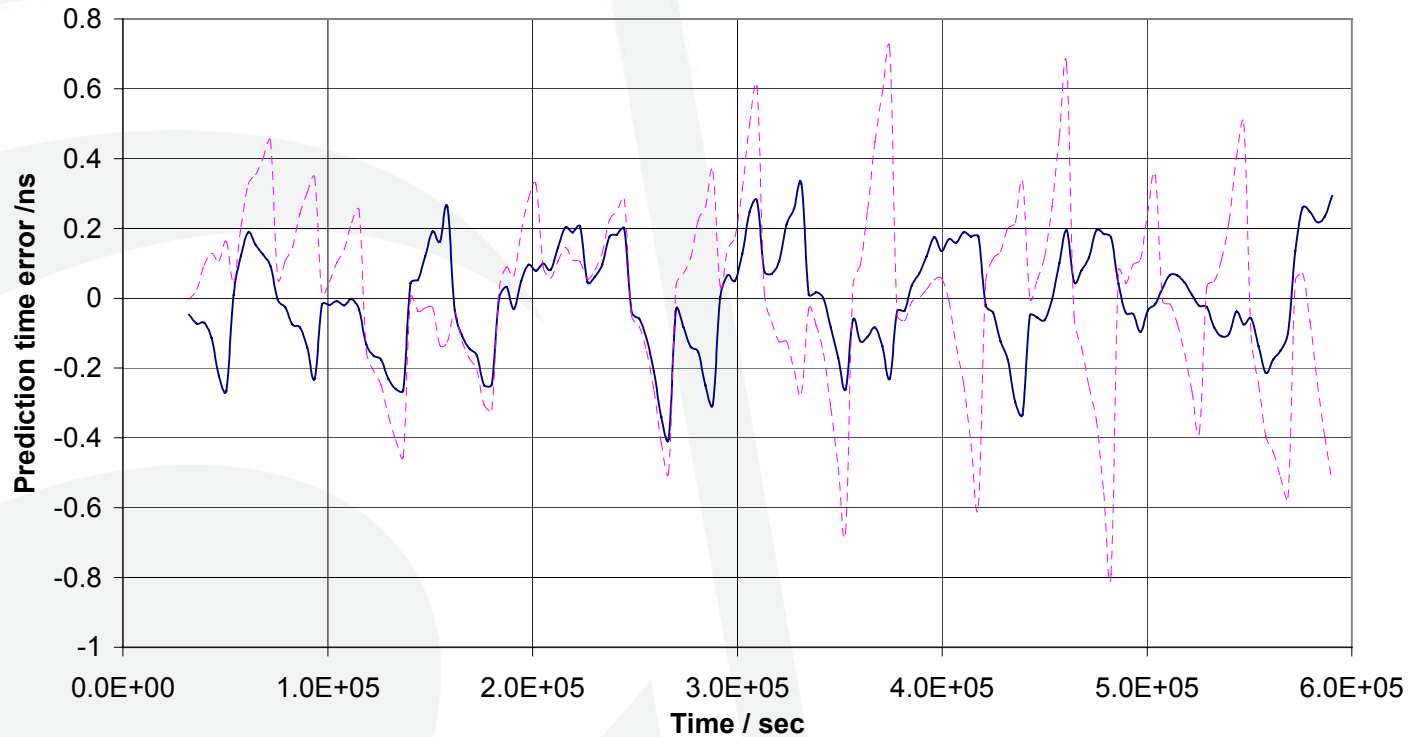


# Prediction Time error for S\_PHM:

**G-S-PHM**  
**T<sub>m</sub>=8h, T<sub>p</sub>=6h**

— Linear    - - - Quadratic

$dT_{RMS}(T_p)=0.226$      $dT_{RMS}(T_p)=0.454$   
 $dT_{RMS}(0-T_p)=0.151$      $dT_{RMS}(0-T_p)=0.271$



T<sub>m</sub> = Measurement Time

Method: Curve fitting during T<sub>m</sub>

Computed by Q. Wang

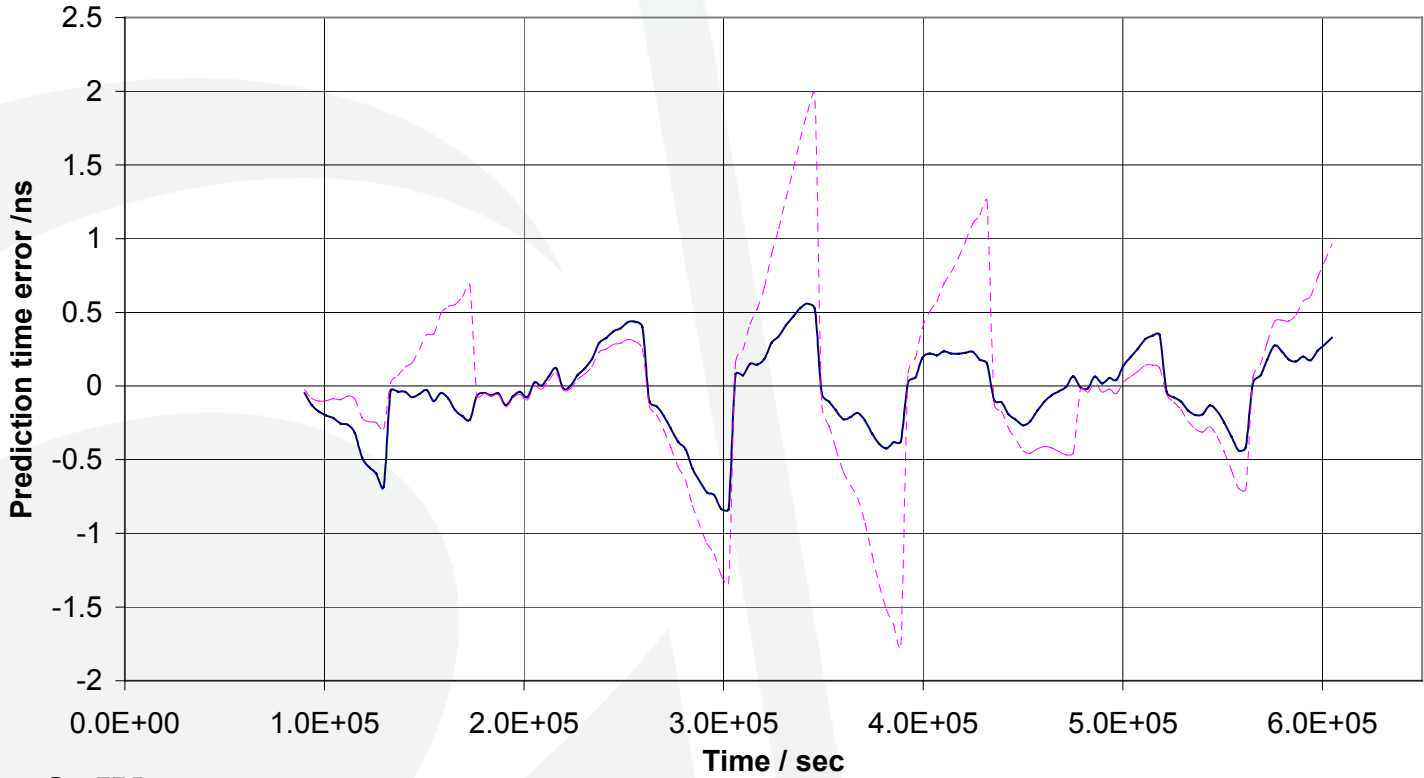
T<sub>p</sub> = Prediction Time

Plot: Diff. Between fit and meas. da

# Prediction Time error for S\_PHM:

**G-S-PHM**  
**T<sub>m</sub>=24h, T<sub>p</sub>=12h**

— Linear    - - - Quadratic  
 $dT_{RMS}(T_p)=0.432$      $dT_{RMS}(T_p)=1.029$   
 $dT_{RMS}(0-T_p)=0.284$      $dT_{RMS}(0-T_p)=0.632$

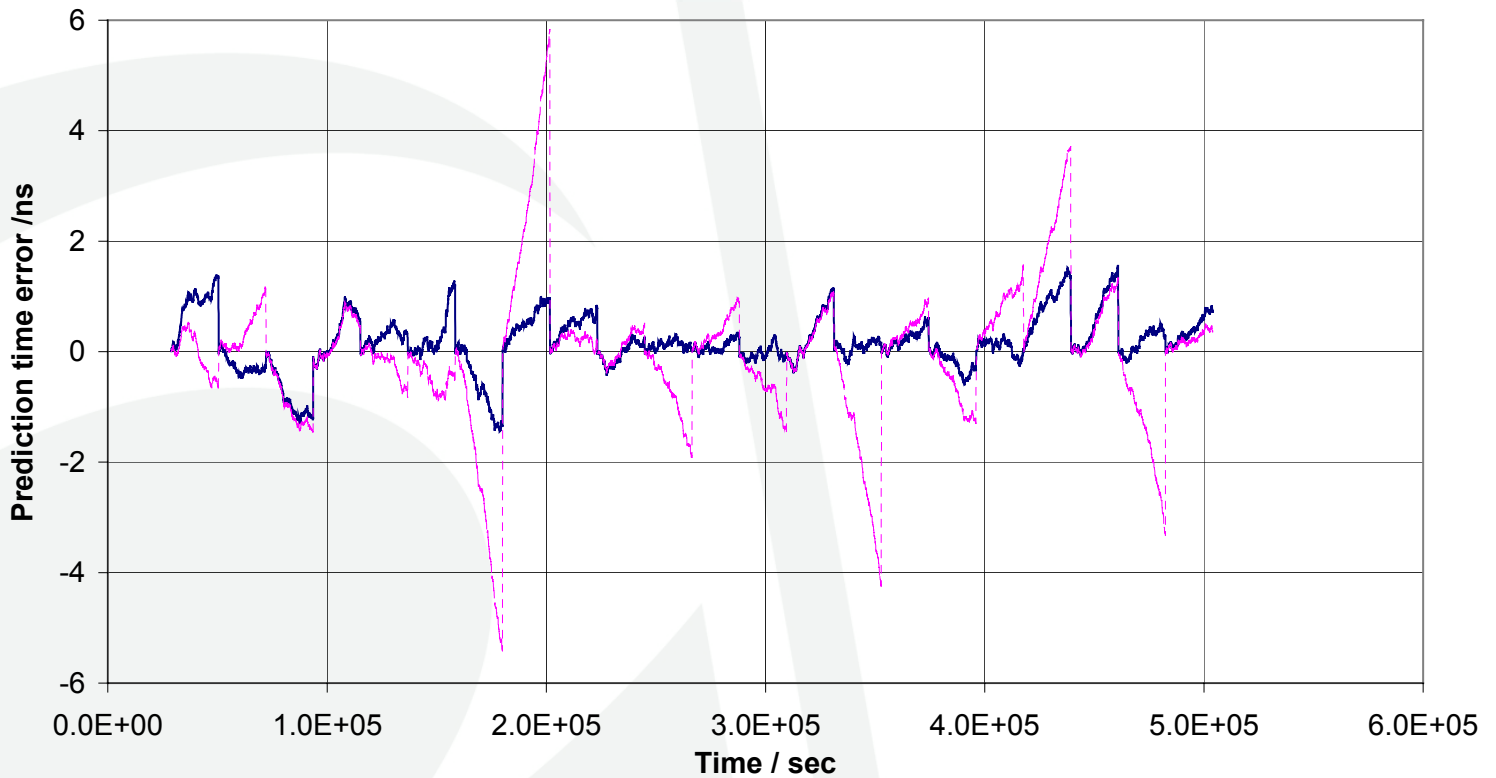


Computed by Q. Wang

# Prediction Time error for RAFS:

**RAFS**  
**T<sub>m</sub>=8h, T<sub>p</sub>=6h**

— Linear - - - Quadratic  
 $dT_{RMS}(T_p)=0.839$     $dT_{RMS}(T_p)=2.396$   
 $dT_{RMS}(0-T_p)=0.519$     $dT_{RMS}(0-T_p)=1.281$



Computed by : Q. Wang

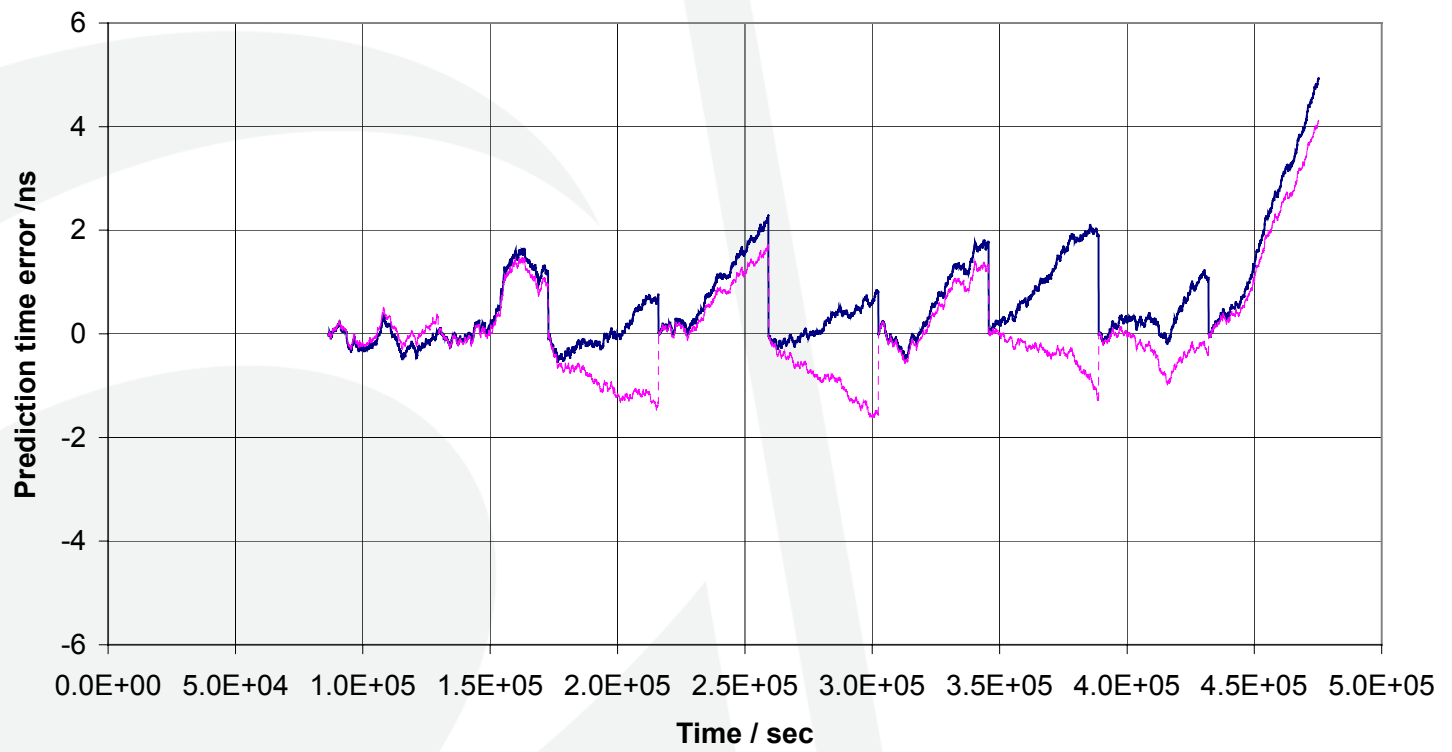
SSOM Engelberg 2007

Pascal Rochat

# Prediction Time error for RAFS

**RAFS**  
**Tm=24h, Tp=12h**

— Linear    - - - Quadratic  
 $dT_{RMS}(Tp)=2.111$      $dT_{RMS}(Tp)=1.783$   
 $dT_{RMS}(0-Tp)=1.139$      $dT_{RMS}(0-Tp)=0.978$

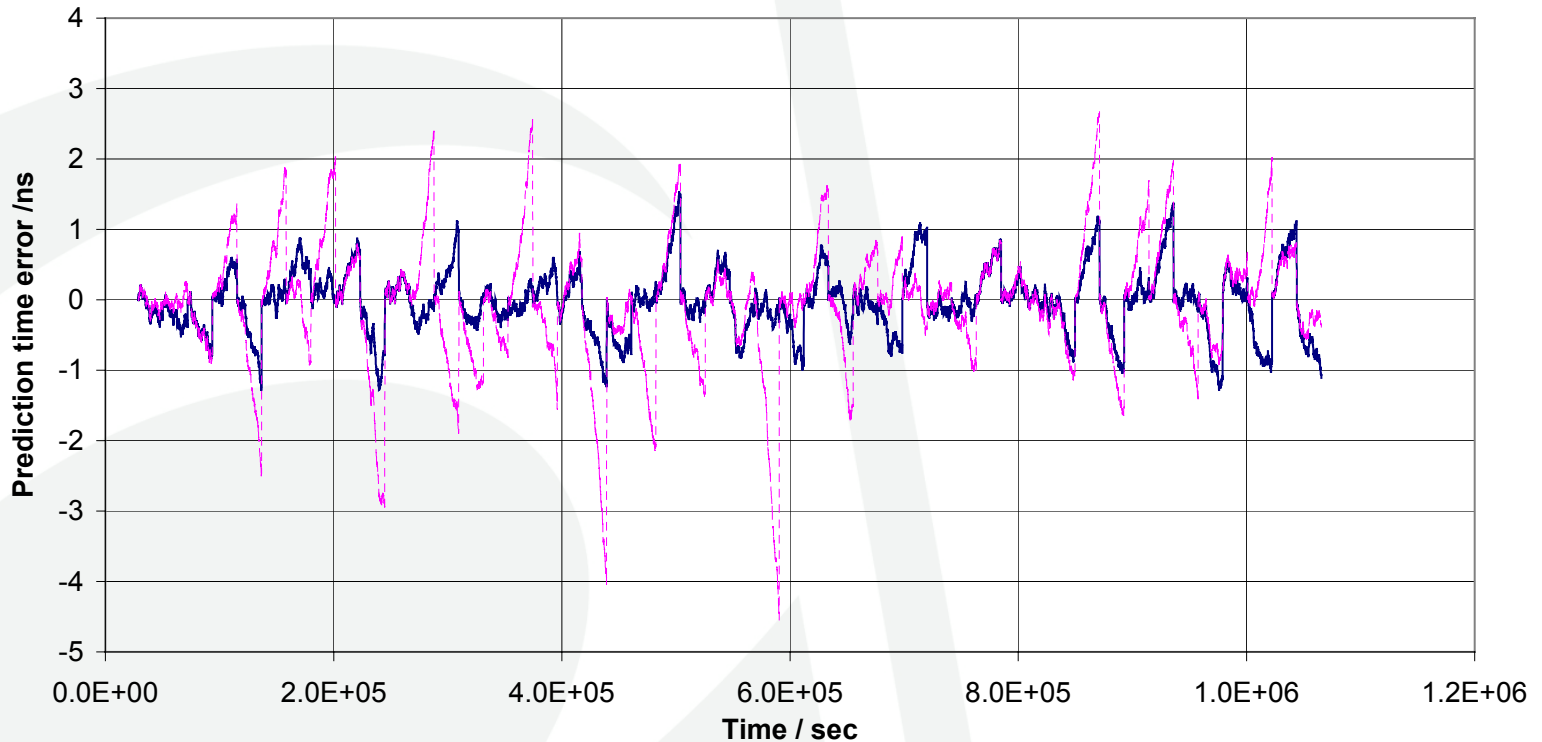


# Prediction Time error for OPCs

**OP-G-Cs**  
**T<sub>m</sub>=8h, T<sub>p</sub>=6h**

— Linear    - - - Quadratic

$dT_{RMS}(T_p)=0.680$      $dT_{RMS}(T_p)=1.618$   
 $dT_{RMS}(0-T_p)=0.453$      $dT_{RMS}(0-T_p)=0.881$

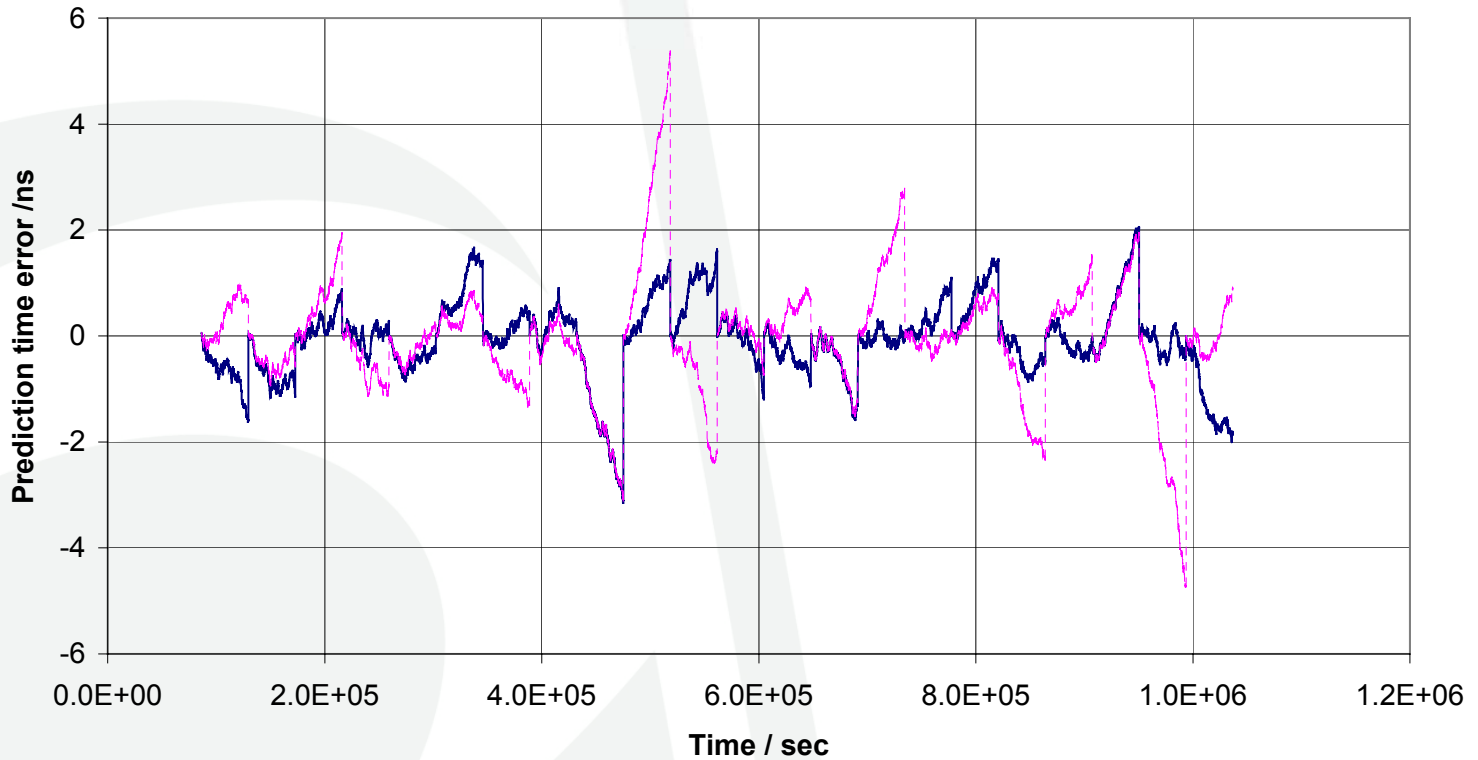


Data collected by TEKELEC SYSTEME on OPCs BB model

# Prediction Time error for OPCs

**OP-G-Cs**  
**T<sub>m</sub>=24h, T<sub>p</sub>=12h**

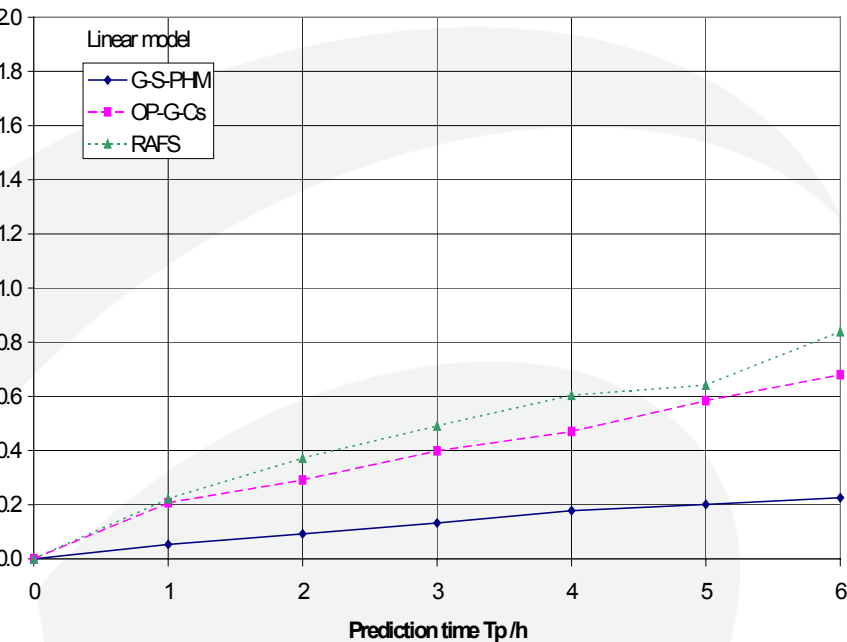
— Linear    - - - Quadratic  
 $dT_{RMS}(T_p)=1.291$      $dT_{RMS}(T_p)=2.089$   
 $dT_{RMS}(0-T_p)=0.750$      $dT_{RMS}(0-T_p)=1.132$



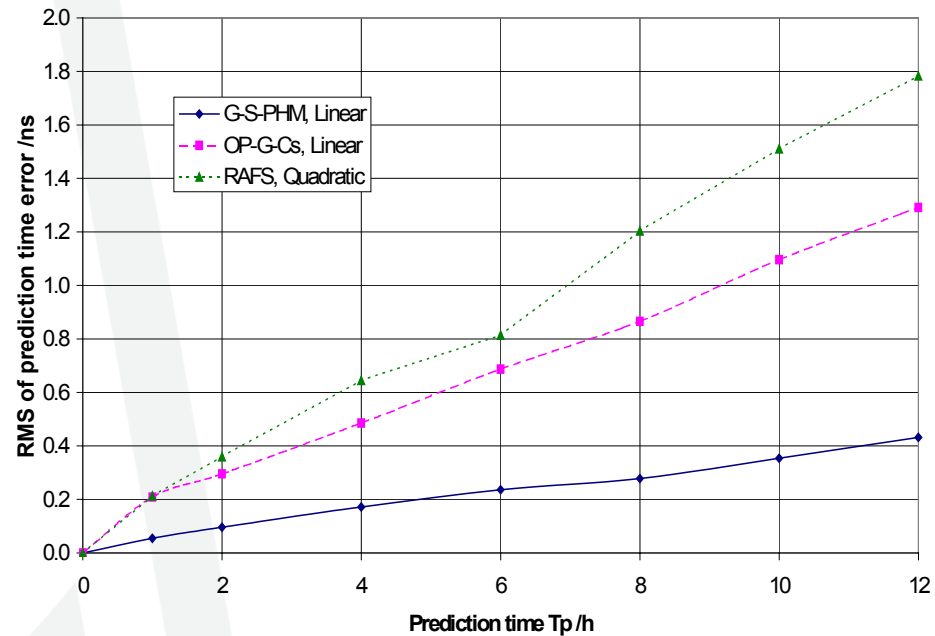
Data collected by TEKELEC SYSTEME on OPCs BB

# RMS of Prediction Time Error:

$T_m=8h$



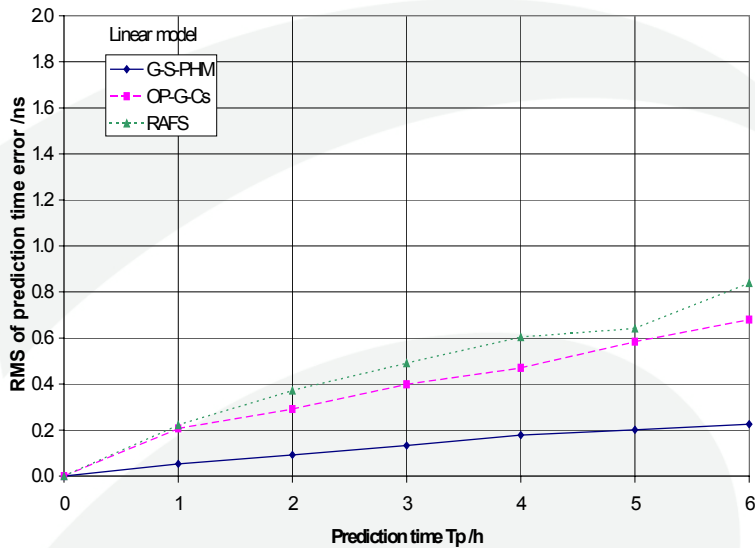
$T_m=24h$



Method : RMS calculation of all error values after  $T_p$

# RMS of Prediction Time Error $T_m=8h$

$T_m=8h$

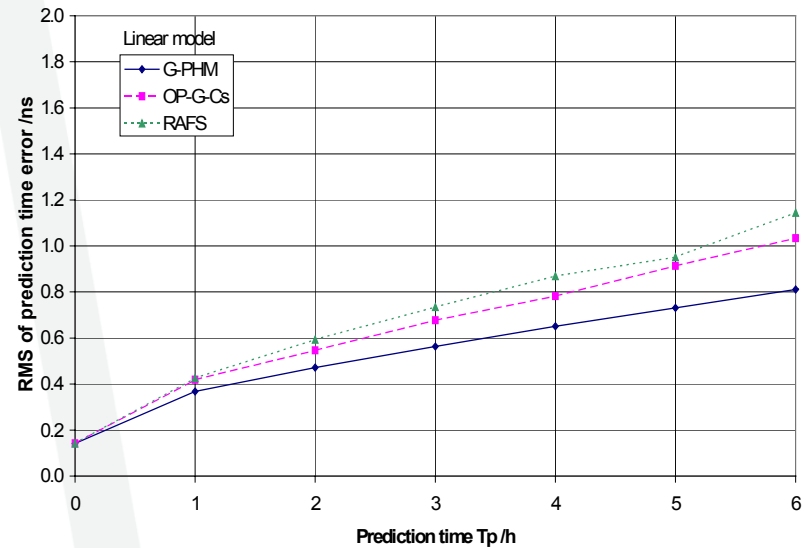


Without geodetic measurement noise

Method : RMS calculation of all error values after  $T_p$

Computed by : Q. Wang / G. Busca

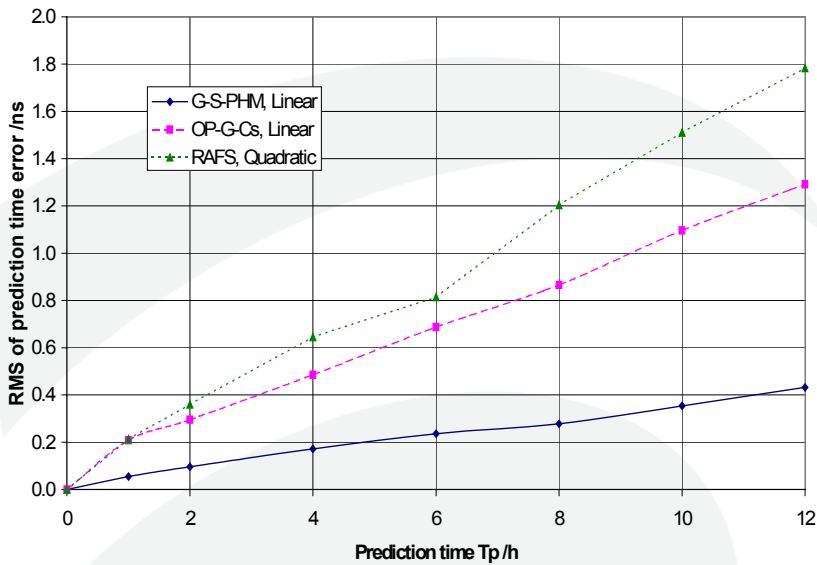
$T_m=8h$ , including the present measurement noise



With present geodetic measurement noise

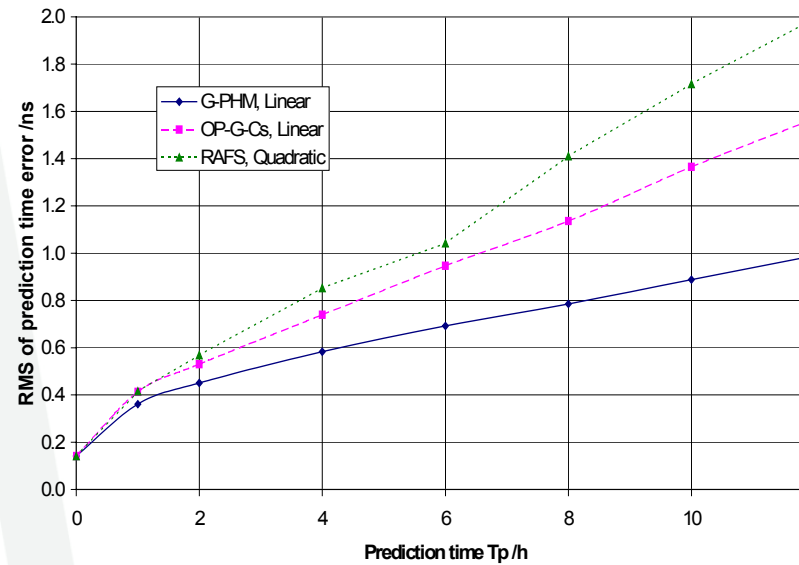
# RMS of Prediction Time Error $T_m=24h$

$T_m=24h$



Without geodetic measurement noise

$T_m=24h$ , including the present measurement system noise



With present geodetic measurement noise

Method : RMS calculation of all error values after  $T_p$

# Conclusions:

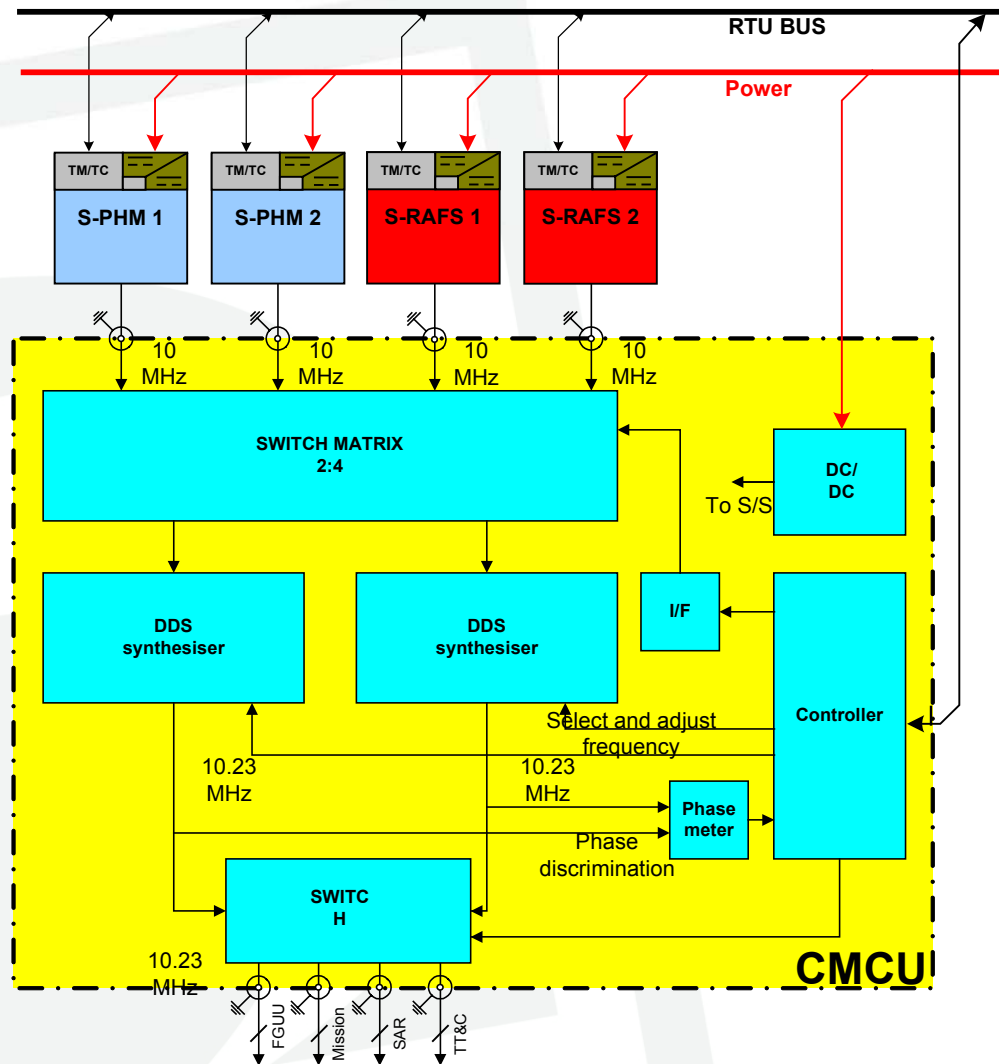
With the present S\_Clocks measurement system noise, and prediction time up to 6 hours , the S\_RAFS is adequate.

For future measurement system noise which are expected to improve by factor of 3 , the S\_PHM will be the more adequate clock for prediction time up to 24 hours.

Navigation accuracy less than 1 m can be achieved with up to hours autonomy with RAFS and up to several days with passive Maser

Cesium clock degrading accuracy while not improving

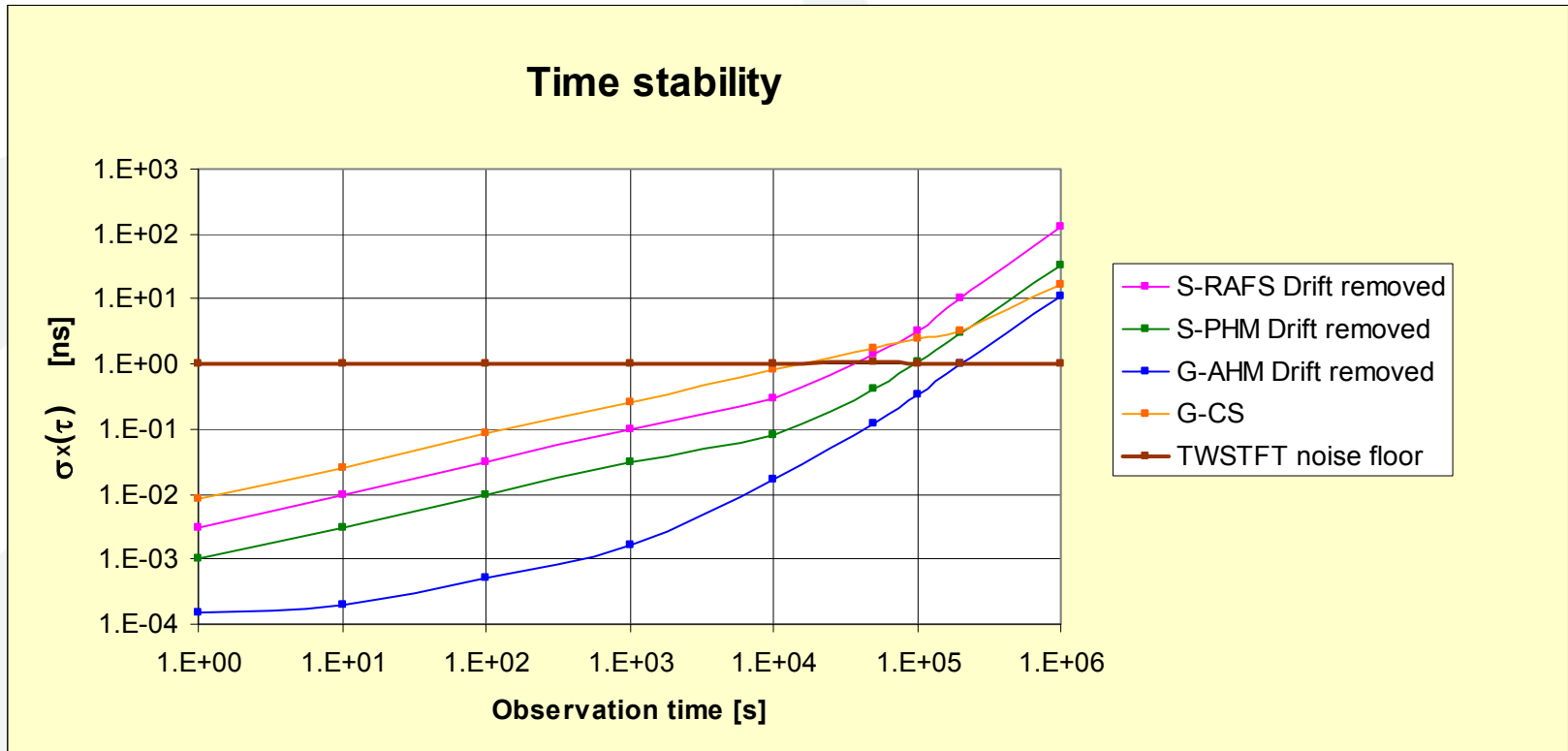
# Galileo On-Board Clocks ensemble



## **Navigation System Time reference generation (ground segment) shall meet following requirements**

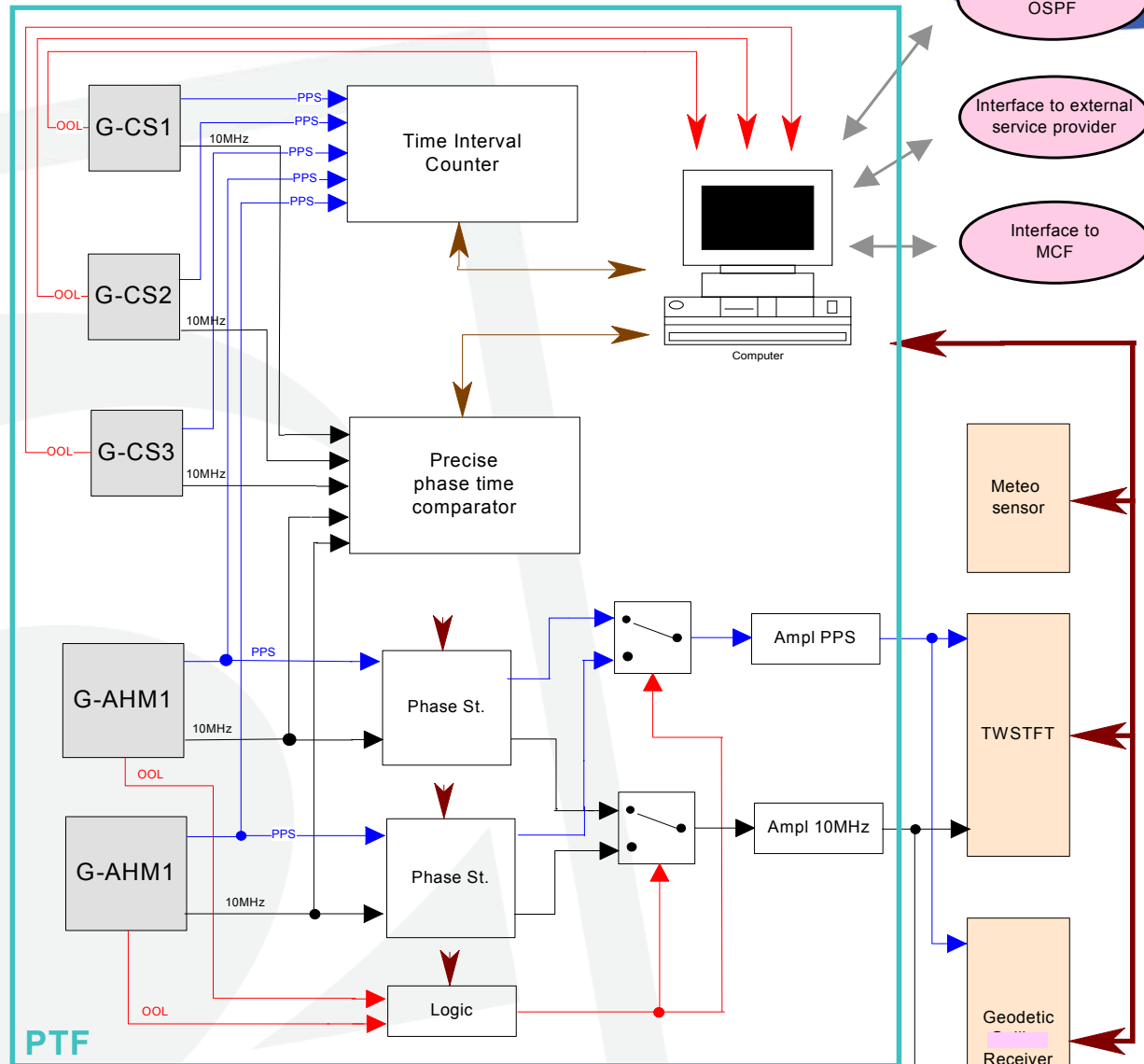
- Be linked to UTC time or other navigation system (s) for interoperability purposes
- Be equipped with both long term & short term stability clocks
- Autonomous switching of redundant master clocks
- Be linked to system by physical generation of time signal to at least one geodetic ground observation station.
- System time shall be generated within 1 ns stability for weeks in order to guarantee proper on board clocks stabilities and set-up on board clocks modelling.

# Space and Ground Clocks Time Stability





# Exemple of System Time generation Architecture



# Conclusions

## Qualified clocks technologies Offering:

1. Metric / sub metric navigation accuracy
2. Up to one week autonomy capability with on board Maser technology
3. Cold + Hot Redondancy on board of each satellite
4. Clocks lifetime expectation of 14 years or more
5. Simple & Mature Technologies offering room for improvements in term of mass & performances
6. Existing groung maser clocks offering better stability than Cesium for inter-operable navigation system linked with precise time transfert equipments



**Thank you very  
much for your  
attention**

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